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Model-predicted geometry variations to compensate material variability in the design of classical guitars

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ABSTRACT

Musical instrument making is often considered a mysterious form of art, its secrets still escaping scientific quantification. There is not yet a formula to make a good instrument, so historical examples are regarded as the pinnacle of the craft. This is the case of Stradivari's violins or Torres guitars that serve as both models and examples to follow. Geometric copies of these instruments are still the preferred way of building new ones, yet reliably making acoustic copies of them remains elusive. One reason for this is that the variability of the wood used for instruments makes for a significant source of uncertainty – no two pieces of wood are the same. In this article, using state-of-the-art methodologies, we show a method for matching the vibrational response of two guitar top plates made with slightly different materials. To validate our method, we build two guitar soundboards: one serving as a reference and the second acting as a copy to which we apply model-predicted geometry variations. The results are twofold. Firstly, we can experimentally validate the predictive capabilities of our numerical model regarding geometry variations. Although applied to guitars here, the methodology can be extended to other instruments, e.g. violins, in a similar fashion. The implications of such a methodology for the craft could be far-reaching by turning instrument-making more into a science than artistic craftsmanship and paving the way to accurately copy historical instruments of a high value.

Introduction

¹ One of the largest problems found in guitar making is the lack of reproducibility: even though people talk of a 'Torres' or a

² 'Hausser' model¹, intrinsic material variations of the wood make every instrument unique. Yet, in their uniqueness, they all ³ share a certain *something* that lets us speak of the different models and their characteristic sounds. To achieve that characteristic

share a certain *something* that lets us speak of the different models and their characteristic sounds. To achieve that characteristic
 sound, luthiers subtly modify the internal bracing of the guitar based on their experience and sensibility. This is where the art
 comes in.

There are, however, many reasons why a more systematic approach to instrument-building is timely and sorely needed. Global warming has already altered the habitat of trees^{2,3}, and tone-wood (the particular kind of spruce used for the soundboard of musical instruments) is bound to become more and more scarcely available. Recent research has clearly shown that the design of a guitar is much more important for the sound than environmental and material variability⁴. What this research lacks, however, is a concrete methodology of how to compensate for those material variations by adjusting the design. In this article, we close this gap by combining the state-of-the-art in parameter identification applied to guitars together with standard optimisation techniques.

The starting point for our research lies with the recent advances in simplified FEM simulations for musical instruments, be they for guitars with model order reduction methods^{5,6} or violins with neural networks⁷. These simplified approaches allow us to obtain the values for the eigenfrequencies of the system in 1/1000th of the time compared to traditional FEM simulations without significant loss of accuracy, allowing us to perform optimisations in a reasonable time^{6,8}. The traditional approach has been used in a variety of musical instruments, from the kantele⁹ to the viola da gamba¹⁰, with numerous studies focusing on string instruments^{4–8,11–17}.

In engineering, shape optimisation using finite element models evolved into a standard method used in many applications^{18, 19}. However, in the field of musical instruments, the only examples that we are aware of are the shape optimisation of a bell, e.g.²⁰, and of vibraphone bars^{21, 22}. To the best of the authors' knowledge, shape optimisation has not yet been applied to string instruments because, in stark contrast to other examples, one does not know how to choose a suitable objective function because one lacks objective criteria to define a 'good' instrument. We can, however, count on the expert knowledge of luthiers and the evaluation of musicians to identify 'good' examples of guitars or violins^{23–25}. Therefore, trying to copy the vibrational



Figure 1. Visualisation of the workflow that we use for matching the modal parameters of two soundboards by compensating the material differences through geometry modifications. Starting from the modal parameters of the reference plate (top 1). By fitting the material parameters of a different plate (top 2) in an initial state with thicker braces than the reference, we can optimise the geometry of the bracing by only looking at the differences in the eigenfrequencies. Finally, the experimental modal analyses show that two different geometries can have a very similar vibrational response.

response of these well-sounding instruments seems to be the most reasonable choice when it comes to designing an objective
 function for the optimisation of string instruments like guitars.

This article's contribution is a methodology for creating a vibrational copy of a reference guitar's top plate. We focus on the 27 top plate since it is not only the most complex but also the most relevant part for sound production in a guitar. Figure 1 shows 28 a diagram of the proposed methodology: Starting from a reference plate (top 1), we identify its modal parameters - namely 29 eigenfrequencies, eigenmodes, and modal damping ratios - in a frequency range up to 1000 Hz. The measured modal parameters 30 of another plate (top 2) with identical geometry besides initially higher braces are fed into a material parameter identification 31 process to obtain a virtual prototype that is able to predict the influence of geometric changes on this plate. This virtual prototype 32 is used in numerical optimisation of the bracing heights to compensate for the differences in the eigenfrequencies between the 33 two plates. The modifications to the bracing height are then applied to top 2 as predicted by the numerical optimisation. Finally, 34 the experimental modal analysis of the reference and the modified copy shows that two different geometries can have a very 35 similar vibrational response. Photographs of the final experimental plates can be seen in Fig. 5. 36 We believe this is an excellent starting point to tackle more general problems regarding the interaction between material 37 and design of musical instruments, as it demonstrates that the traditional geometrical reconstruction of the outer shape of 38

and design of musical instruments, as it demonstrates that the traditional geometrical reconstruction of the outer shape of
 instruments is not necessarily the best to obtain a standard vibrational response. Our approach shows that noticeable geometric
 modifications must be considered if a reference plate's vibrational response is to be achieved.

41 Results

We constructed two guitar top plates according to a simplified Torres fan bracing pattern¹. Great care was taken in building them as similarly as possible by matching the density of the braces used and the final weight of the top plates.

A diagram of the two plates and their bracing height indicated by colours can be seen in Fig. 1, for 'top 1 - reference' and top 2 - initial'. We have coloured the top plates in different shades to indicate that they have different material parameters. In

⁴⁶ particular for top 1 the identified longitudinal stiffness is $E_L^1 = 11.6$ GPa and the density is $\rho^1 = 403$ kg m⁻³ and for top 2 we



Figure 2. (a) Comparison of eigenfrequencies calculated from the numerical model and experimentally identified eigenfrequencies for the initial configuration of top 1. (b) Comparison of eigenmodes between the numerical model and the experimental modal analysis. (c) Conducted bracing height changes to validate the numerical model. (d) Relative frequency changes caused by the bracing height variation in the numerical model and the experiment.

get $E_L^2 = 9.31$ GPa and $\rho^2 = 407$ kg m⁻³. In total, 35 material parameters are identified for each top as all braces are handled individually (see Methods). We have started from very thick braces (7mm in the fan region) for two reasons. First, it serves us as a validation of the numerical model for different geometric configurations. Secondly, when we try to optimise top 2 to fit the vibrational response of top 1, we need to have a range of possible heights for the braces, and since taking out wood is easier than adding it, we decided to start from an oversized configuration.

Figure 2a compares the eigenfrequencies of the experimental setup and the numerical model once the material parameters 52 of the braces and the plate have been identified for top 1. The experimentally identified eigenfrequencies are the mean values 53 from 4 measurements, with the eigenfrequencies varying in a mean range of $\pm 0.7\%$ for the first 24 modes. Figure 2b shows the 54 modal assurance criterion matrix²⁶ for the first 24 modes of the top plate — showing that the modal similarity is excellent. 55 In order to validate the model, we implement the modal identification in two stages. Starting from the initial configuration 56 with rather thick braces (Fig. 2c top 1 - initial), we arrive at the reference configuration (Fig. 2c top 1 - reference) and compute 57 the relative change of the eigenfrequencies in the numerical model as well as in the experimental setup. Figure 2d shows the 58 relative change in the frequencies for the experiments and simulations. Notice that the material parameter identification is only 59 made for the initial configuration of top 1. 60

Once the material parameters are fully identified, we can develop a linear model of the relative influence of the bracing height on the eigenfrequencies (see Methods). Figure 3a shows the normalised influence of the braces' heights on the eigenfrequencies of the first 6 modes of the top plate. The vertical inset shows the modal shape associated with each eigenfrequency, whereas the horizontal diagrams show which of the braces it refers to. Notice that for symmetry, we only take pairs of braces in the fan, except for the central one. In Fig. 3b, the height changes necessary to influence the eigenfrequency of modes 1 and 2 are depicted vividly.

In order to quantify the variability in the eigenfrequencies due to material or design parameter changes, we sample the parameter space for each of them in the range of possible values for each (see Methods). We sample the eigenfrequencies



Figure 3. (a) Correlation matrix between eigenfrequencies of the first modes and the height of specific bracing areas. (b) Bracing height changes that are necessary to change the eigenfrequency of the specified modes. (c) Relative standard deviation of eigenfrequencies caused by feasible bracing height changes and possible material parameter changes within spruce wood.

10000 times and obtain a distribution that turns out to be very close to normal, similarly to the results reported in⁷. From those 69 distributions, and for each eigenfrequency, we measure the standard deviation of the frequency change. Figure 3c shows the 70 relative standard deviation in percentage for the first 10 modes of the top plate. Interestingly, the variation due to material 71 parameters is, on average, slightly larger than that due to geometric variations, which explains why the wood selection is such a 72 critical step in instrument making. Particularly modes 1, 2, 4, 6, and 8 are extremely sensitive to the variation in the material 73 parameters and are much less affected by the bracing height. We conclude that these modes depend more on the plate's specific 74 stiffness than on the braces' geometry. Nevertheless, these results explain that our method to reproduce the vibrational response 75 works well since we started with very similar woods that only vary in their stiffness and had almost the same density. 76

Finally, in order to find out the necessary bracing heights of top 2 so that it vibrates as top 1, we use a straightforward optimisation process. Instead of using directly the finite element simulations at each iteration step, we use the linear model of Fig. 3a to solve the optimisation problem of the eigenfrequencies of both top plates with the objective function

$$\mathscr{L} = \sum_{m=1}^{13} \left(\frac{f_m^{\text{ref}} - f_m}{f_m^{\text{ref}}} \right)^2.$$
⁽¹⁾

To minimise this objective function, we use Matlab's *fmincon* constrained optimisation algorithm for the seven brace heights shown in Fig. 3a.

The optimisation problem converges after around 100 iterations, and the height profile for the braces is carved into top 2 as depicted in Fig. 4a. After the optimisation, the difference in eigenfrequencies between top 1 and top 2 can be observed for the initial and modified configuration (purple and green line in Fig. 4b respectively), as well as the difference predicted from the numerical model (red line). The mean error between top 1 and top 2 is less than 2 %, which is extremely well predicted by the

86 numerical model.



Figure 4. (a) Conducted bracing height changes given by the optimisation to fit the soundboard copy to the reference. (b) Experimentally identified relative frequency difference compared to the reference soundboard before and after the height changes. (c) Modal damping ratios of the reference and the copy soundboard in comparison. (d) One exemplary experimentally identified mode in comparison between reference and copy.

Even though our objective function only considers the eigenfrequencies, the damping ratios and modal shapes are also well 87 fitted, showing that the choice of the objective function is very well suited for the problem. Figure 4c shows the damping ratio 88 for the reference top 1 and the optimised top 2. The match is very good for the first six modes, which are the most radiative 89 ones, so we expect a similar tonal performance of both top plates. The fit of the modal shapes is even better, showing an average 90 MAC of 0.92 for the first 13 modes. An example of the mode similarity can be seen in Fig. 4d where for mode 5, a MAC of 91 0.98 is achieved. In conclusion, by fitting the first 13 eigenfrequencies, our model can correctly predict the height profile that 92 top 2 needs to yield a similar vibrational response to top 1. The results are even great for both modal shapes and damping ratios, 93 which come 'for free' when optimising the eigenfrequencies. In Fig. 5 the plates are depicted in their final state. The height 94 differences between the harmonic bars are clearly visible while the small height differences in the fan are barely recognisable. 95

96 Discussions

In this article, we have presented a methodology for the material identification of a guitar top plate and a predictive framework 97 that allows one to 'copy' a target vibrational response. The framework uses state-of-the-art methodologies to rapidly compute 98 a linear model of the vibrational response of the plate as a function of the bracing height. We then use that linear model to 99 minimise an objective function based on the first 13 eigenfrequencies to obtain the geometry that matches the vibrational 100 response in a top plate of a different material than the original. We carve the optimised bracing height profile in our experimental 101 top and show that, indeed, the results are very similar, not only for the eigenfrequencies but also for the damping ratios and 102 the modal shapes of the first 13 modes. Note, however, that the damping ratios cannot yet be explicitly controlled with the 103 methodology since they are not included in the objective function. 104

Thanks to the model order reduction technique, we can sample a vast array of material and geometrical values and study how geometry and material parameters affect the eigenfrequencies. Previous results on the use of metamaterials for thin wooden plates²⁷ show that, to a certain degree, the density and stiffness of the plates can be purposefully controlled. By matching the density and the longitudinal stiffness of the top and the back plate, we can be certain that the modifications of the braces will be minimal⁶.



loops for experiment mouting

Figure 5. Pictures of the two plates in their final configuration with depicted bracing heights next to the braces. Especially for the harmonic bars, the height difference between the corresponding braces of the two plates is clearly visible.

The implications of these results are far-reaching in the field of instrument making: By quantifying how close we can come 110 to a desired vibrational response with a given material, we have taken guesswork out of the equation in guitar-making. One 111 could argue that the equipment used in this research is far beyond the reach of standard contemporary instrument-making 112 workshops. However, the principles behind our method can be applied with any setup of modal identification. Some luthiers 113 already perform this as part of their workflow to characterise their instruments^{28,29}. Furthermore, recent advances in the 114 development of efficient surrogate models like neural network prediction of the vibrational response of wood^{7,30} and parametric 115 model order reduction for shape optimisation³¹ make us hopeful that in the near future this optimisation method can become a 116 fundamental part of guitar-makers' toolboxes. 117

Finally, the ability to deliberately achieve a certain vibrational response is unheard of in classical instrument making. Instead 118 of blindly following older designs or searching for new ones based only on intuition, we propose a method for scientifically 119 and methodically producing a copy of a given instrument. This is not only relevant in guitar-making but in other instruments 120 as well. A case in point is violins, where some historical instruments more than 300 years old are no longer suited for actual 121 playing. This method could help us hear instruments that are no longer playable (and the example of Stradivari's Messiah 122 comes immediately to mind) and create accurate acoustic copies of them. Whether the accuracy of the method is sufficient to 123 produce indistinguishable acoustic copies of instruments still remains an open question in the field of musical perception^{25, 32, 33}. 124 125 The path is rather long still, but this is a necessary step in the direction of turning instrument making more into a science than a mysterious art. 126

127 Methods

128 Guitar plates construction

The guitar plates were built with same-grade wood pieces bought from the same dealer (Rivolta Wood, Desio, Italy). The wood used is Abete Rosso (*Picea abies*). Each guitar plate is made of two bookmatched pieces supplied by the dealer. The plates were glued up with fish glue (Kremer pigments, Aichstetten, Germany) using the traditional guitar-making methods. Six weeks passed between glueing the top plate and glueing the braces to allow the wood to regain its original moisture content.

The wood for the braces was sorted by density, and the two top plates were matched as closely as possible to have the same 133 weight distribution on space. The mass variation of the braces for either guitar was less than 3 gr. Due to the size of the brace 134 wood, an independent material parameter identification could not be used, and we opted for a bulk characterisation from the 135 MOR model. The braces were glued to the soundboard with rabbit glue (Cremona tools, Cremona, Italy). The braces were 136 planed by hand to a standard cross-section of 7x7 mm and glued in a fan pattern based on a simplified Torres model from 1884¹, 137 the Stradivari of guitar-making. Harmonic bars had a 23.5x7 mm cross-section. Before glueing the fan bars, we measured their 138 density and ordered them in such a way that the heaviest bars were in the centre of the top plate and the lightest on the sides. 139 The impact of the glue was in no way characterised and assumed irrelevant. After construction, the plates were shipped from 140 Cremona to Stuttgart, where they were kept in a climate-controlled room for 3 months before starting the measurements. 141

142 Experimental modal analysis

Experimental modal analysis is the standard method when it comes to identifying modal parameters of vibration structures 143 and has been applied to various musical instruments^{5,29,34,35}. A setup with the guitar plates being suspended by very soft 144 springs approximating free boundary conditions was developed for the experimental modal analysis. The plates were kept, and 145 measurements were taken in a climate-controlled room, with a relative humidity of $55 \pm 1\%$ and a constant temperature of 146 24°C. One can find details on the climate-controlled room in³⁶. The plates' velocities were measured with a Polytec PSV-500 147 Scanning Laser Doppler Vibrometer, and an automatic impulse hammer acts as the excitation device³⁷. The excitation with 148 the hammer yielded a reproducible excitation of frequencies up to 1000 Hz, and the maximum forces on the guitar plates 149 did not exceed 3.0 N. A total of 220 mobilities were measured. These measurements were composed of 110 points on the 150 soundboard, where the velocity is measured and two distinct excitation positions between the fan braces. Each measurement 151 resulted in a data sequence of duration T = 0.8 s. Using a sampling step of $\Delta t = 1.6 \cdot 10^{-4}$ s, the width of each frequency bin of 152 the corresponding Fourier transform was $\Delta f = 1.25$ Hz. Longer measurements would have resulted in zero padding due to the 153 faded signal and were, therefore, avoided. The complex mode indicator function, in combination with enhanced frequency 154 response functions, was used to identify the modal parameters of the plates. Details on the method can be found in³⁸, and a 155 detailed description of the application to a classical guitar is included in⁵. The uncertainty of the modal parameter changes, 156 given in the error bars in Figs. 2 and 4, was calculated from 15 measurements throughout the modification process of top 1 by 157 interval arithmetic³⁹. 158

159 Material parameter identification

The material parameter identification follows the approach described in detail in⁶. Detailed finite element models of the guitar plates act as the key pieces of the approach^{40,41}. The models were created in the commercial software Abaqus with free boundary conditions and an orthotropic material model for all the braces⁴². The plates were discretised with linear shell elements of Abaqus type S4, while the braces' discretisation was carried out with linear C3D8 volume elements. In former publications, rigid tie constraints have shown to be a reasonable assumption for binding the plate and the braces together^{5,6}. Hence, this approach was used in this publication, too. The degrees of freedom of the full-order model with a very fine discretisation sum up to N = 400128.

Unfortunately, the detailed model takes too much computational time to be evaluated thousands of times during the 167 parameter identification procedure. Furthermore, the parameter space would contain up to 107 material parameters if all braces 168 with all their material parameters were to be identified individually. Thus, a projection-based Krylov approach for parametric 169 model order reduction was applied to reduce the number of degrees of freedom in an efficient surrogate model to n = 600 while 170 keeping a good approximation of the full-order model's results up to 1000 Hz. This was reached by matching the transfer 171 function of 4 inputs and outputs distributed over the plate at 20 frequency shifts equally distributed in the frequency range and 172 60 parameter expansion points created with a Sobol sequence as explained in⁶. General information on model order reduction 173 can be found in^{43,44} while a review of parametric model order reduction techniques is given in⁴⁵, and the used software is 174 described in⁴⁶. The order-reduction approach reduces the computational time to calculate the first 30 modal parameters from 175 78 s with the full-order model to 0.04 s with the reduced-order model, corresponding to a numerical speedup of 1950. On 176 a set of test data consisting of 100 evaluations for the first 30 eigenfrequencies, the common coefficient of determination is 177 $R^2 = 0.96$ between the reduced-order model and the full-order model⁴⁷. 178

In the reduced-order model, parameter dependency is preserved for the 35 most influential parameters chosen with the help of a sensitivity analysis. The parameters kept for the plates are the density ρ , the Young's moduli $E_{\rm L}$ in longitudinal and $E_{\rm T}$ in the tangential direction, and the shear modulus $G_{\rm LT}$. The parameters are identified for each brace individually. However, the number of parameters varies between the different braces as follows. Parameters for the fan braces comprise ρ , $E_{\rm L}$, and the shear modulus $G_{\rm LR}$ with the subscript *R* denoting the radial direction. The higher braces directly above and below the soundhole are characterised by ρ , $E_{\rm L}$, the Young's modulus in radial direction $E_{\rm R}$, and $G_{\rm LR}$. The horizontal brace on the upper part of the plates is parameterised with ρ and $E_{\rm L}$. Values for spruce from⁴⁸ are used for all the remaining values.

The parameter identification procedure follows a two-step approach. Firstly, the 35-dimensional parameter space were explored as done in⁶ using a sampling approach based on a Sobol-sequence and one million samples⁴⁹. In the second step, an objective function comparing eigenfrequencies and eigenmodes was evaluated, and the eight best-performing solutions were given into the Matlab *fmincon* algorithm as starting values. In both sets, constraints were set in such a way that the total mass of the plates would not allow variations beyond $\pm 5\%$ with respect to the experimentally measured value. As a second set of constraints, the material parameter values were not allowed to exceed bounds taken from literature^{4,48,50,51}. The best solution with respect to the objective function evaluated for the first 24 modes of the algorithm was used for the geometry optimisation.

Geometry optimisation

¹⁹⁴ The finite element model with the identified material parameters served as a virtual prototype to apply the changes to the braces.

¹⁹⁵ The model of top 1 was used for validation purposes, as shown in Fig. 2, while the model with the material parameters for top 2

was used to optimise the bracing heights of top 2 to match the modal parameters of top 1, as visible in Fig. 4. In this procedure, 196 seven independent height parameters were used for the ten braces as the symmetry in the fan braces was kept constant. Again, 197 the full-order model turned out to be unsuitable for optimisation purposes due to its high computational cost. For this reason, a 198 linear regression model was fitted for the correlation between the first 13 eigenfrequencies and the heights of the braces. The 199 linear regression model was trained with a set of 950 parameter samples of bracing heights created from a Sobol sequence in a 200 realistic range of $h_{\text{range}} = [1 \text{ mm}, 7 \text{ mm}]$ for the lower braces and $H_{\text{range}} = [8 \text{ mm}, 23.5 \text{ mm}]$ for the harmonic bars. On a set of 201 test data consisting of 50 further samples, the coefficient of determination is $R^2 = 0.96$. 202 This regression model was then used in the optimisation process to identify the optimal bracing heights. Since the finite 203

²⁰³ This regression model was then used in the optimisation process to identify the optimal bracing heights. Since the finite ²⁰⁴ element model approximates the absolute values of the eigenfrequencies with a small systematic error, the desired relative ²⁰⁵ change of eigenfrequencies was used as optimisation criterion as depicted in the violet curve in Fig. 4b. Hence, the systematic ²⁰⁶ error between the finite element model and the experiment did not influence the results. Then, the mean squared error of the ²⁰⁷ first 13 eigenfrequencies was minimised using Matlab's *fmincon*. The only constraints set in the optimisation process were the ²⁰⁸ lower and upper bounds, and they were specified as written above in h_{range} and H_{range} .

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212 Author contributions statement

A.B.: Conceptualization, Methodology, Software, Data Curation, Writing- Original draft preparation, Visualisation, Investi gation; S.G.: Plate Construction, Conceptualization, Writing- Original draft preparation, Visualisation, Investigation; M.V.:
 Software, Methodology; F.A.: Supervision, Funding Acquisition, Writing- Reviewing and Editing A.S.: Supervision, Funding
 Acquisition, Writing- Reviewing and Editing; P.Z.: Supervision, Funding Acquisition, Writing- Reviewing and Editing; P.E.:

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218 Additional information

219 Competing interests

²²⁰ The authors declare no competing interests.

221 Data availability statement

The datasets generated during and/or analysed during the current study are available from the corresponding author on reasonable request.

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