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Avoiding the “Great Filter”: A Projected Timeframe for Human Expansion Off-World

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Abstract

A foundational model has been developed based on trends built from empirical data of space exploration and computing power through the first six plus decades of the Space Age which projects earliest possible launch dates for human-crewed missions from cis-lunar space to selected Solar System and interstellar destinations. The model uses computational power, expressed as transistors per microprocessor, as a key broadly limiting factor for deep space missions’ reach and complexity. The goal of this analysis is to provide a projected timeframe for humanity to become a multi-world species through off-world colonization, and in so doing all but guarantees the long-term survival of the human race from natural and human-caused calamities that could befall life on Earth. Beginning with the development and deployment of the first nuclear weapons near the end of World War II, humanity entered a ‘Window of Peril’ which will not be safely closed until robust off-world colonies become a reality. Our findings suggest the first human-crewed missions to land on Mars, selected Asteroid Belt objects, and selected moons of Jupiter and Saturn can occur before the end of the 21st century. Launches of human-crewed interstellar missions to exoplanet destinations within roughly 40 lightyears of the Solar System are seen as possible during the 23rd century and launch of intragalactic missions by the end of the 24th century. An aggressive and sustained space exploration program, which includes colonization, is thus seen as critical to the long-term survival of the human race.

1. Introduction

“I now am become as Death, the destroyer of worlds” (Voices of the Manhattan Project, 2017). With those words, recalled from his readings of the *Bhagavad Gita*, Dr. J. Robert Oppenheimer, chief scientist of the Manhattan Project, announced that on the morning of July 16, 1945 humanity attained the power of global scale self-destruction. As the people of the world would come to know by the atomic bombs dropped on Hiroshima and Nagasaki less than a month later, ending humanity’s most destructive war, the age of self-annihilation had indeed dawned. In the decades to follow world powers built up their arsenals of nuclear weapons and added biological weapons, a new dimension of mass lethality. Under such existential threats, now for most of a century and with the added pressures of sharp population increase, strained resources, and climate change, it

is easily understandable how one could become convinced humanity is living through an ever-expanding debris field of our own troubled history. Is it just in our nature that we are bound to hurdle towards self-annihilation once the means are at hand, falling victim to the cosmic “Great Filter” (Hanson, 1998)? The term “Great Filter” is a conceptualized probability threshold that could pose a barrier to the evolution of intelligent life.

Consideration of that question, which has challenged researchers since before the dawn of the Space Age, manifests analytically as the “L” factor (lifespan of technical/communicative civilizations) in the famed Drake Equation (Drake, 1965) and more recently found by statistical modeling to be the Equation’s most influential term (Cai, 2020). Indeed, “Our transient existence has lasted for less than 10 one-billionths of cosmic history so far on a tiny rock we call Earth, surrounded by a vast lifeless space” (Loeb, 2020). Against this dark backdrop, however, we are reminded that, “We are more than just our genes” (Hawking, 2018). Human minds are sufficiently complex to render unlivable our home world have also reached across the vast gulf of space to touch other worlds. Given the fact the Earth faces natural threats such as asteroid impacts, supernovae radiation, and super-volcanic activity, “our chance for survival could improve if some people choose to move away from Earth. Currently, all our eggs are in one basket. Venturing into space offers the advantage of preserving our civilization from a single-planet disaster” (Loeb, 2021). In recent decades, as robotic missions throughout the Solar System progressed, much speculation has been given to the next steps including colonization (Schmidt, 1996).

Focusing on the assurance for humanity’s survival that would be afforded by permanent, self-sustaining, and genetically diverse colonization off-world, and determining when such a Herculean task can be expected to occur is paramount. Estimates for how long a technological civilization might last if confined to just its home world varies widely, from the pessimistic limit of destruction occurring in the immediate future, to “a few centuries but not much longer” (Loeb, 2021) to approaching 8,000 years (Cooper, 2013). Such a broad range is to be expected given the complexities inherent in individual and group human behaviors integrated across the Earth’s nearly 8 billion people. Estimates for off-world colonization’s timeframe can be expected to vary as well, functionally dependent as they are on the rate of technological development, government and private industry priorities, and public support, among other factors. However, more than 60 years since the Space Age have brought rapid development of humanity’s off-world capability, from Sputnik in 1957 to men on the moon in 1969 to robotic probes having visited the far reaches of the Solar System and Kuiper Belt. Hence, a key parameter within humanity’s control for impacting the odds of long-term survival can be found in how many years it will take to establish at least one permanent, self-sustaining and genetically diverse off-world colony. This ‘window of peril’, which opened in 1945 for the technological civilization of Earth, will remain so until such colonies become a reality.

But how do we meet the challenge of estimating how long until robust off-world colonies are established? Answering, even in the very approximate, such a far-reaching query requires extensive modeling involving both quantitative and qualitative data from the start of the Space Age to the present, plus projections into the coming decades of the 21st century that encompass an array of unknowns such as findings from the Moon, Mars, Jovian and Saturnine satellites already underway in this decade. Rather than trying from the outset to build a complex multi-variable model, the first steps of an incremental approach is presented here whose intent is to lay

the foundation for building the complex model by first keying on a single critical parameter. The choice for this critical parameter is computing power, and is driven by the fact of its own technological development running parallel to that of space exploration technology. Cutting edge space exploration technology historically relies on computational capability, ease of quantifying, and rapid progress. For the purposes of this analysis, computing power will be measured as the number of transistors that can be contained within a single microprocessor. The logic of this choice follows directly from the clear dependence of scientists, inventors, engineers, medical researchers and others on their ability to gather, process and interpret the vast quantities of information required to conceive, design, build, test, operate and maintain the new technologies required for increasingly complex and distant human missions into deep space. “[Shawn] Domagal-Goldman and [Giada] Arney, [NASA Goddard astrobiologists] envisage future exoplanet missions where AI technologies embedded on spacecraft are smart enough to make real-time decisions, saving the many hours necessary to communicate with scientist on Earth” (Shekhtman, 2019).

At this point it should be noted that computing power expressed as transistors per microprocessor does have an analogy to the human brain. More specifically, the human brain contains $\sim 10^{11}$ neurons and $\sim 10^{14}$ synaptic connections among those neurons (Zimmer, 2011). Further, “A transistor can be thought of as an analog to the synapse (connection) between two neurons” (Whitwam, 2013). As well, cutting edge technology such as using ‘persistent photoconductivity’ found in some perovskite semiconductors as a kind of ‘optical memory’, this “...phenomenon can also mimic synapses in the brain that are used to store memories” (National Renewable Energy Laboratory, 2021). Hence, computing power as described by transistor count per microprocessor can be thought of as a cousin to the human brain at the basic structural level, with both organic and synthetic computational devices being used in concert to develop the new technologies of the Space Age.

Finally, it should be noted that any estimate of deep space mission timing merely projects a date of earliest launch from Earth (or, more generally, cis-lunar space, which means "not beyond the moon" in Latin), but does not include actual travel time to a given mission's final destination or the follow-on time interval to establish human colonies once the first crewed landing is made. On this note, while chemically-powered rockets can likely suffice for missions to Mars, Jovian space and, perhaps, as far as the Saturn system, their inherent exhaust velocity limitation of about 5 km/sec precludes their consideration for practical interstellar travel (Benford, 2013). Travel to interstellar destinations, while still in the realm of speculative fiction, can possibly be met with highly advanced technology such as generational star ships and/or breakthrough propulsion systems – e.g., Bussard Ramjet (Bussard, 1960) utilizing a CNO-catalyzed nuclear fusion cycle (Whitmire, 1975; Benford, 2013). In any case, greater computing power than is available today will be needed to engineer, build and operate such designs.

2. Methodology

As computing power grows at an ever-accelerating rate, approximately doubling every two years during much of the second half of the 20th century (Moore, 1975) and continuing into the 21st century, so too has humanity's reach into outer space. The robotic component of the Space Age, which began with Sputnik in 1957, quickly advanced to the first lunar flybys in 1959 (Luna 1 and 2), flyby missions to Venus (Mariner 2, 1962), Mars (Mariner 4, 1964), a successful soft

landing on the Moon (Luna 9, 1966), and launches of later successful flybys of the outer planets by Pioneer 10 in 1972, Pioneer 11 in 1973 and Voyager 1 and 2 in 1977. The model discussed below is based on the confluence of these two exponential trends: computing power (expressed as transistors per microprocessor) as a function of time and distance (measured in astronomical units, AUs) from launch point Earth to the first robotic deep space missions' farthest successfully achieved objective as a function of time.

2.1 Computational Power and Deep Space Missions Over Time

As will be seen, both trends can be easily plotted in a semi-log manner with the linear horizontal axis being the number of years since the start of the Space Age (Figure 1a). Exponential trend curves are then best-fit to the plotted data, yielding:

$$C = 15.70226 \exp(0.34821 T) \quad (1)$$

where C = computing power, T = time in number of years since 1957, and the correlation coefficient $R^2 = 0.996$. See Figure 1b for the graphical representation of data corresponding to equation (1).

And:

$$D = 0.04846 \exp(0.31272 T) \quad (2)$$

where D = distance from launch point Earth to farthest successfully achieved mission objective in AUs (minimal distance from Earth's orbit to objective's orbit), T = time in number of years since 1957, and the correlation coefficient $R^2 = 0.989$. See Figure 1b for the graphical representation of data corresponding to equation (2). Please refer to Figure 2 at the end of this section for a simplified flow chart of the development of this and other relationships which comprise the core of our model.

With the commonly time dependent equations of these two data sets in hand, it is then a straight forward exercise to combine the two equations so as to eliminate T , their time dimension, creating a new power relationship between computing power and distance from launch point Earth to farthest successfully achieved robotic mission objective:

$$C = 457.2437 D^{1.11334} \quad (3)$$

where the correlation coefficient $R^2 = 0.999$. It should be noted that the computing power vs. time data set contains 24 data points over the 61 years between 1957 and 2017 (Roser, 2013) while the deep space missions vs. time data set includes only 5 data points. This limitation, implicit in its reliance on the relatively small number of deep space missions' empirical data, cannot be avoided.

Equation 3 provides a relationship between how far distance-wise robotic deep space missions have progressed relative to concurrent computing power. This relation a fundamental element which underlies the technology and procedures necessary for venturing into deep space for extended periods of time, and ultimately establishing off-world colonies, is coupled to the physical reach of such efforts.

Figure 1: Left-panel; Computational power as expressed in number of transistors per microprocessor from 1971 to 2017, time expressed as number of years since start of the Space Age (1957), used in empirically derived equation (1) (Roser, 2013); Right-panel: First of their kind robotic deep space missions' (flybys) planetary objective minimum distance from Earth orbit (in astronomical units) versus number of years from start of Space Age (1957) to missions' launch year, used in empirically derived equation (2).

While it may appear more straightforward to estimate the timing of additional and more extensive deep space missions by basing such projections solely on selected successful missions already flown – i.e., simply following the established trend further out in space and time, doing so would blunt the influence in any such estimate from the critical component of concurrent technological development. Indeed, when just the simple logarithmic extrapolation of first successful flyby missions actually flown is used as the basis – i.e., equation (2), solved for time as a function of distance, to estimate timing of far-reaching human-crewed missions, an ever-widening discrepancy arises with respect to increasing mission objective distance. As an example, for the nearest interstellar mission to Proxima Centauri, the aforementioned simplistic extrapolation predicts a first successful launch date of 2080 for human-crewed mission, 174 years *sooner* than predicted by our model which takes into account as well the projected time required to develop mission-necessary computing power. When one considers Moore's Law is expected to soon to run up against material and engineering limitations (Waldrop, 2016), this while radical new forms of computing to carry out ever more complex calculations (e.g., quantum computing) are yet to be perfected, the means to accomplish the developmental dimension of humanity's most daunting endeavors cannot be ignored.

2.2 From Robotic Missions to Human-Crewed Landings

Robotic missions have and will continue to pave the way for human crews to follow. However, the leap from relatively inexpensive (and ultimately expendable) machine voyagers to craft capable of keeping groups of highly complex living organisms alive in the most hostile of environments is an enormous leap in nearly every design and construction detail. As one aspect, humans evolved under the protection of Earth's strong magnetic field and dense atmosphere which work in combination to screen-out the vast majority of high energy particles from

interstellar space – commonly referred to as cosmic rays. Direct exposure to cosmic rays, even for relatively short periods of time, would pose grave health risks to an unprotected crew. Although shielding can be employed to meet this need, this would add substantially to the engineering requirements of crewed spacecraft along with additional mass necessitating more powerful engines, which in turn adds further to the mass of the vehicle presenting engineers with what amounts to a Newtonian conundrum. Nonetheless, the first 10 years of the Space Age brought a single spectacular example of bridging robotic to human landings, this beginning with Luna 1 and 2's flyby of the Moon in 1959, Luna 9's soft landing in 1966, and culminating in 1969 with the first crewed landing by Apollo 11. While this progression provides a critical data point for this analysis, at least one other data point would be needed to begin proposing a quantifiable time-distance relationship between robotic and human missions. It is at this point a degree of speculation is needed concerning humanity's next likely target for 'boots on the regolith' – Mars. Mariner 4 first flew by Mars in 1964 and was followed by the first successful soft landers, Viking 1 and 2, in 1976. While humans have yet to step foot on the Martian surface, a first crewed landing in the 2030s is envisioned (Ahmed, 2021). For the purposes of this analysis, we will venture a somewhat conservative assumption that such a landing successfully launches in 2038. When going from launch of a first actual or first projected flyby mission to a first robotic lander mission for a given destination, a fixed number of years is simply added to the actual or projected flyby launch date which corresponds to the actual delta realized for Mars, that of 11 years (Mariner 4's flyby in 1964 to Viking 1 and 2's launch in 1975). The implicit assumption is that the technological leaps required to go from robotic flybys to robotic landings are minor compared to going from robotic flyby to human landings. Further, for robotic missions the intermediate step between flybys and landings, that of an orbiter without a lander, is ignored per the leap from flyby to orbiter also being minor when compared to the challenges of placing a complex device safely on the surface of another world. Finally, while several of the Soviet era Venera probes to Venus did successfully place robotic landers on the surface, the severe conditions characteristic of Venus greatly limited their operating window and would almost certainly rule out Venus as a viable destination for human landings, much less colonization, in the foreseeable future. For these reasons only Mariner 2's Venus flyby in 1962 is included in the construction of our model.

With two sets of data points for time interval and distance when going from robotic flyby to human landing missions, noting again that the second set of data points, this involving Mars, are partially speculative, it is possible to propose a mathematical relationship. While any curve can be perfectly fitted to this small a data set, this analysis will confine itself to a logarithmic relationship between time (expressed as number of years between first successful robotic flyby and human landing) vs. distance from launch point Earth to mission objective (measured in AUs). The resulting equation is as follows:

$$T_{fh} = 12.824 \ln(D) + 86.532 \quad (4)$$

where T_{fh} = number of years from launch of first successful robotic flyby mission to launch of first successful human landing. In the case of the Moon, $D = 0.00256$ AU while T_{fh} is 10 years. For Mars, $D = 0.376$ AU (closest orbital approach, not necessarily the distance a crewed or non-crewed vehicle would take in the instance of a Hohmann transfer orbit) while T_{fh} is projected to be 74 years.

The rationale for using a logarithmic relationship is founded in the nature of technological development itself, specifically the familiar S-curve of development vs. time. First, there is a relatively slow run-up over decades which can be regarded here as our going from the Wright Brothers' first powered flight of a heavier than air vehicle in 1903 to Sputnik in 1957 – i.e., the lower portion of the S-curve. Technological progress then experiences a sharp upward acceleration over a short period of time, this covering the first ~20 years of the Space Age, that took us from Sputnik to Voyager 1 and 2 – i.e., the steep middle portion of the S-curve. Finally, with much of the 'low hanging fruit' of technical barriers having been surmounted for relatively near-Earth mission objectives, the long and gently curving upper section of the S-curve is encountered. It is the upper half of the S-curve that is relevant to our analysis and best expressed mathematically in the logarithmic form of equation (4) and as illustrated below in Figure 2.

Figure 2: Number of years separating launch of first successful robotic flyby to robotic landing and first successful robotic landing to human landing, for Moon and Mars missions. Note: Launch of first successful human landing on Mars is assumed to occur in calendar year 2038.

Finally, there is consideration for the time interval in going from the first crewed landing to the first self-sustaining, genetically viable (i.e., robust) colony. As no such colony has yet been established off-world, actual data for folding into this analysis is pending though the coming years with the prospect for the beginnings of said colonies (NASA, 2019). For the purposes of this analysis, it will be assumed that once the planned Artemis Base Camp is successfully established near the Moon's south polar region, launches for true colonization can begin within a relatively short time frame from the successful first crewed landings' launch to Mars and beyond.

2.3 Assembling the Model

Figure 3 outlines how the above empirically derived equations are combined to create the model for extended deep space exploration. It must again be emphasized that computing power, while

a broadly affecting factor, is only one such factor in a highly complex interplay of dynamic influences on how far and how soon humanity will establish off-world colonies. As such, this model is best considered a starting point version of more rigorous versions to come (see 5. Future

Work).

Figure 3: Empirical equations 1 and 2, in combination with distance to mission objective and flyby mission launch year data, yield projections for extended missions' launch years and computational power required. Note: "f", "f'", "g", "h" and "i" denote functional relationships.

3. Results and Discussion

The data and derived equations contained herein estimate the projected calendar year dates and associated computational power that will be required for the successful launch of initial robotic landing and human landing missions from Earth (effectively cis-lunar space) to a given set of deep space mission objectives.

3.1 Actual and projected timing for deep space missions and associated computational power

In Table 1, the Solar System mission destinations were chosen per their history of robotic visits and in the case of the Moon, the human landings that have taken place. Note again the assumption for first human landing on Mars to launch from Earth/cis-lunar space in 2038. The rationale for the interstellar mission destinations are as follows:

Proxima Centauri – The closest known star to the Solar System at 4.2 lightyears, this active red dwarf is strongly suspected of having a least two planets and at least one of those, Proxima Centauri b, in the star's habitable zone (Anglada-Escudé, 2016).

Tau Ceti – The closest known G-class (Sol-like) star not part of a tandem star system, it is 11.9 lightyears away and hosts at least four planets and at least one in the star’s habitable zone (<http://phl.upr.edu/press-releases/twonearbyhabitableworlds>).

Trappist System – A cool red dwarf star 40.5 lightyears away that is thought to host at least seven terrestrial-sized planets and several among those potentially in the star’s habitable zone (Gillon, 2017).

4 kpc from Center of the Milky Way – An annular region located approximately 4,000 parsecs from the center of the Milky Way Galaxy (closest point: ~14,000 lightyears from the Solar System, in the direction of Sagittarius and the galactic center) has been suggested by statistical modeling as the region of the Milky Way most likely to have contained complex life, including technological civilizations, in the past (Cai; Jiang, 2020).

Utilizing the data in Table 1, Figure 4 provides a graphical representation of when, in terms of Common Era calendar year, the first successful launches from Earth/cis-lunar space have occurred and are estimated to occur for both robotic and human landings to interplanetary and interstellar mission destinations.

Table 1: Actual and projected timing of first successful robotic and human missions for selected destinations within the Solar System and interstellar space.

Tables for LookUp Data on Pre-Chosen Destinations and Projections for Earliest Possible Mission Launch Dates									
Destination	Moon	Mars	Asteroid Belt	Jovian System	Saturn System	Proxima Centauri	Tau Ceti	Trappist System	4 kpc from CoMW
Dist. from Earth (AUs)	0.0026	0.3763	1.5587	3.9501	8.0412	265,486	752,526	2,562,570	882,424,035
1st Flyby	1959	1964	1972	1972	1973	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched
1st Robotic Lander	1966	1975	Not Yet Launched	Not Yet Launched	1997	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched
1st Human Landing	1969	2038	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched	Not Yet Launched
Actual and Calculated Projections for Calendar Year of Launch from Earth of First Successful Mission Type and Destination and Corresponding Computational Power Requirement									
Launch Year of 1st Successful Flyby Mission	1959	1964	1972	1972	1973	2007	2010	2014	2032
Year Computational Power First Achieved	1948	1964	1968	1971	1973				
Launch Year of 1st Successful Robotic Landing	1959	1975	1979	1982	1984	2018	2021	2025	2043
Year Computational Power First Achieved	1959	1975	1979	1982	1984	2018	2021	2025	2043
Launch Year of 1st Successful Human Landing	1969	2038	2064	2076	2086	2254	2270	2290	2383
Year Computational Power First Achieved	1969	2038	2064	2076	2086	2253	2269	2289	2381
Launch Year of 1st Successful Human Landing	1969	2038	2059	2073	2097	2243	2258	2277	2365
Year Computational Power First Achieved (1)	1969	2038	2059	2073	2097	2242	2257	2276	2363

Utilizing the data in Table 1, Figure 4 provides a graphical representation of when, in terms of Common Era calendar year, the first successful launches from Earth/cis-lunar space have occurred and are estimated to occur for both robotic and human landings to interplanetary and interstellar mission destinations.

Figure 4: Graphical representation of data in Table 1 for launch date from Earth/cis-lunar space of robotic and human landing missions having destinations ranging from the Moon to approximately 14,000 lightyears from the Solar System.

3.2 Interpretation of selected missions' results

The above results are calculated for nine pre-chosen mission destinations, the first five of which are within the Solar System and representative of actual mission destinations flown starting in the late 1950s while the remaining four destinations are interstellar and run the range from our nearest stellar neighbor to approximately halfway to the center of the Milky Way Galaxy. Continued is a brief discussion for each destination:

Moon: The world closest to Earth and, so far, the only world humans traveled to and landed upon. Having already been visited by humans, this places the Moon somewhat removed from the intent of this analysis given much of the most challenging aspects of its colonization have either been solved or are now coming into reach in the very near term with NASA's proposed 'Artemis Base Camp' by the late 2020s (NASA, 2019). Given that the Moon is only a four-day trip from LEO by existing spacecraft technology, resupply from Earth would be far easier than even the next closest world for colonization, Mars, which currently takes a minimum of roughly six months to reach.

Mars: As stated in Section 2.2, an assumption has been made for a first successful human landing on Mars to occur in the late 2030s, this likely coming as an extension of the Artemis Program (NASA, 2019). Hence, Mars is effectively treated as a fixed point in our analysis with speculation that at least the beginnings of self-sustained colonization can be expected well before the end of the 21st century. Supporting this assumption is analysis suggesting Mars to be the most economically viable location in the Solar System for colonization (Zubrin, 2007) and the focus of at least one well-funded private venture.

Asteroid Belt: Popular media references to human missions to the Asteroid Belt which lay between the orbits of Mars and Jupiter often involve mining operations. This is not without some scientific backing, such as the potential for substantial metals recovery from 16 Psyche which is the target of a NASA robotic orbiter mission set to launch in August 2022 (JPL/NASA, 2019). Our model predicts first human launch to an Asteroid Belt object could happen by the mid-2060s, with speculation that any human colonies to follow may eventually take hold on some of the larger bodies such as Ceres, 2 Pallas or 4 Vesta.

Jovian System: Our model predicts a first human-crewed launch to Jovian space may be possible by the mid-2070s. A more optimistic assessment was assembled by NASA in a 2003 study which detailed a human-crewed launch to the outermost Galilean moon, Callisto, by the mid-2040s (Troutman; et al, 2003). While the four Galilean moons Io, Europa, Ganymede, and Callisto are the most obvious targets for robotic exploration, only the outermost moon, Callisto, is sufficiently removed from Jupiter's intense radiation fields to consider human exploration and possible colonization within the 21st century. Further robotic explorations of the Jovian system are planned such as the European Space Agency's Jupiter Icy Moons Explorer, set to launch in 2022, and NASA's Europa Clipper, scheduled to launch in 2026 - the findings from which may well further the case for humans to follow or suggest other, more viable, destinations.

Saturn System: Saturn and its many moons, while roughly twice as distant from Earth as the Jovian system, is projected to be reachable via a first human-crewed launch that would take place in the mid-2080s. Saturn's largest moon, Titan (possessing of a thick atmosphere of light hydrocarbons and the only interplanetary moon to have been visited thus far by a robotic lander), is an obvious target for long range plans owing largely to its vast hydrocarbon resources. However, Enceladus is at least as intriguing for human colonization due to the subsurface ocean it is suspected of hosting, as indicated by observations of recurrent H₂O plumes erupting from its south polar region (Brown; et al, 2014). While the Jovian moon Europa is also strongly suspected of having a subsurface ocean, the aforementioned radiation concerns may make Enceladus a superior target for human colonization.

Interstellar Destinations: As discussed in Section 3.1, first crewed launch to even the nearest interstellar destination, the Proxima system, is not envisioned until the mid-23rd century. Once, however, this great leap has been sent on its way, launches to star systems 'moderately' more distant such as Tau Ceti at nearly 12 lightyears away are projected to be possible within another two to three decades. A first human launch to more distant solar systems such as Trappist at approximately 40 lightyears from the Solar System would entail another roughly two decades. Finally, the launch of a human-crewed voyage on the galactic scale, such as traversing the Milky Way's disk to reach halfway to its center, remains a goal much more distant in time and would not be expected possible before the late 24th century.

It must again be emphasized that all projections for human-crewed missions described above are first possible launch dates from Earth/cis-lunar space. Without revolutionary breakthroughs in propulsive engineering, human voyages – and any colonization that would follow – to interstellar destinations is highly unlikely to be undertaken. Fortunately, the Solar System presents a target rich interplanetary environment for astronauts to further explore and adventuring colonists to conquer starting here in the 21st century.

4. Conclusions

A simple model based on trends built from empirical data of space exploration and computing power through the first six plus decades of the Space Age has been developed which projects earliest possible launch dates of human-crewed missions to Solar System and interstellar destinations. The model algebraically folds in the expected broadly limiting factor of computational power, as quantitatively expressed in transistors per microprocessor, to the demonstrated extent of robotic and human missions to the Earth's Moon and the other planets and moons of the Solar System. While space exploration for purely scientific ends is a noble goal, and certainly worthy in and of itself of the costs and sacrifices of such great endeavors, the goal of this analysis is to provide a timeframe for humanity to become a multi-world species through off-world colonization that would logically follow on the heels of earlier human landings. By creating at least one, and preferably multiple, self-sustaining and genetically viable off-world colonies, humanity can best assure its long-term survival from both natural and human-caused calamities that threaten life on Earth. The immediacy of such threats began with the development of the first crude nuclear weapons and has proliferated in multiple directions in the three quarters of a century since the end of World War II. In the early 21st century we find ourselves in the midst of a 'Window of Peril' which, having been opened in 1945, will not safely close until said colonies become a reality.

Per the findings of our model, launches from cis-lunar space of the first human-crewed missions to land on Mars, selected minor planet/planetoid bodies in the Asteroid Belt, at least one of the four Galilean moons of Jupiter, and at least one of the moons of Saturn should be possible before the end of the 21st century – if computational power is the only limiting factor and advances in microcircuitry continue apace. Human-crewed missions to interstellar destinations, while theoretically possible, will be dependent not only on development of onboard AI capability but also technologically advanced propulsive systems, the development of which is logically dependent as well on an ever-greater ability for computation. If undertaken and pursued to success, humanity will have not only all but assured its own survival but will have become a true spacefaring civilization (Zubrin, 1999). Off-world colonization, like the Great Filter's self-annihilation, is within humanity's reach - which we chose as our future will constitute Earth's reply to Fermi's Paradox.

5. Future Work

The methodology for deriving the projected results of this analysis suggests at least two avenues for further expansion. Referring to Figure 5 (a generalization of Figure 3), below, they are:

1. Folding-in additional 'Associated Empirical Functions' (i.e., other affecting factors for equation (1) such as the trend of annual inflation-adjusted public and private sector funding for space exploration programs) to the 'Primary Empirical Function' of actual missions flown, expressed above as equation (2). This would yield a new version of the 'Modified Primary Empirical Function' for equation (3) through which the new 'Associated Empirical Function' is expressed in a revised 'Objective Function'. In addition to the funding parameter, other empirically derived associated functions may include but are not limited to international cooperation (building on the historic successes of early

robotic missions to Venus and Mars, Apollo-Soyuz, and the International Space Station), public support, prior robotic survey findings, political and social priorities. Note that while some of these parameters may prove difficult to quantify in the direct sense, usable trends for the purpose of this type of analysis may still be extracted from data covering the six plus decades of the Space Age. From here, multiple 'Objective Functions' would be generated that would then need to be either algebraically combined to a 'Global Objective Function' or, more simply, their individual 'Projected Results' consolidated via weighted averaging scenario.

2. Adding an iterative feature to the calculations in which initial and intermediate model results are sequentially compared against the data set for the 'Primary Empirical Function'. Any discrepancies above a chosen tolerance would then generate successive correction factors that would be folded into the 'Objective Function' for each iteration until 'Projected Results' are brought within the chosen tolerance.

Figure 5: Methodology for combining an empirically derived primary function with an empirically derived associated function (future work item 1), plus optional recursive loop for successive corrections (future work item 2).

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Data Statement

The data underlying this article are available in the article and in its online supplementary material. For additional questions regarding the data sharing, please contact the corresponding author at Jonathan.H.Liang@jpl.nasa.gov.

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